

**LSTC**  
Livermore Software  
Technology Corp.

**DYNA**  
MORE SWISS

## New Features for Metal Forming in LS-DYNA

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# Outline

- Definition of material hardening behavior
- Best-fit for comparison of geometrical deviations
- Improvements for One-Step simulations
- Uniform mesh refinement of selected region
- Mesh fusion
- Adaptivity and Trimming for sandwich sheets
- Miscellaneous

pre

post

solver

adaptivity

misc



**Definition of material  
hardening behavior**

# Definition of hardening behavior

- Currently different material models offer different options to define yield curve

**\*MAT\_037**

HLCID Load curve ID expressing effective yield stress as a function of effective plastic strain in uniaxial tension.

**\*MAT\_036**

HR Hardening rule:  
EQ.1.0: linear (default),  
EQ.2.0: exponential (Swift)

**\*MAT\_133**

HARD Hardening law:  
EQ.1.0: Exponential hardening:  $\sigma_y = k(\varepsilon_0 + \varepsilon_p)^n$   
EQ.2.0: Voce hardening:  $\sigma_y = a - be^{-c\varepsilon_p}$   
EQ.3.0: Hansel hardening  
EQ.4.0: Gosh hardening:  $\sigma_y = k(\varepsilon_0 + \varepsilon_p)^n - p$   
EQ.5.0: Hockett-Sherby hardening:  $\sigma_y = a - be^{-c\varepsilon_p^q}$   
LT.0.0: Absolute value defines load curve ID or table ID with yield stress as functions of plastic strain and in the latter case also plastic strain rate.

# New keyword: \*DEFINE\_CURVE\_STRESS

- Defines yield curve based on commonly used hardening laws.
- Weighted combinations of hardening laws are possible.
- \*DEFINE\_CURVE\_STRESS can be used by any material model that accepts a curve to define the hardening behavior.

Define second card with the same LCID if ITYPE = 11.

Card 1	1	2	3	4	5	6	7	8
Variable	LCID	ITYPE	P1	P2	P3	P4	P5	
Type	I	I	F	F	F	F	F	
Default	none	none	none	ITYPE 1: 0.001	none	none	none	

# New keyword: \*DEFINE\_CURVE\_STRESS

Implemented hardening laws:

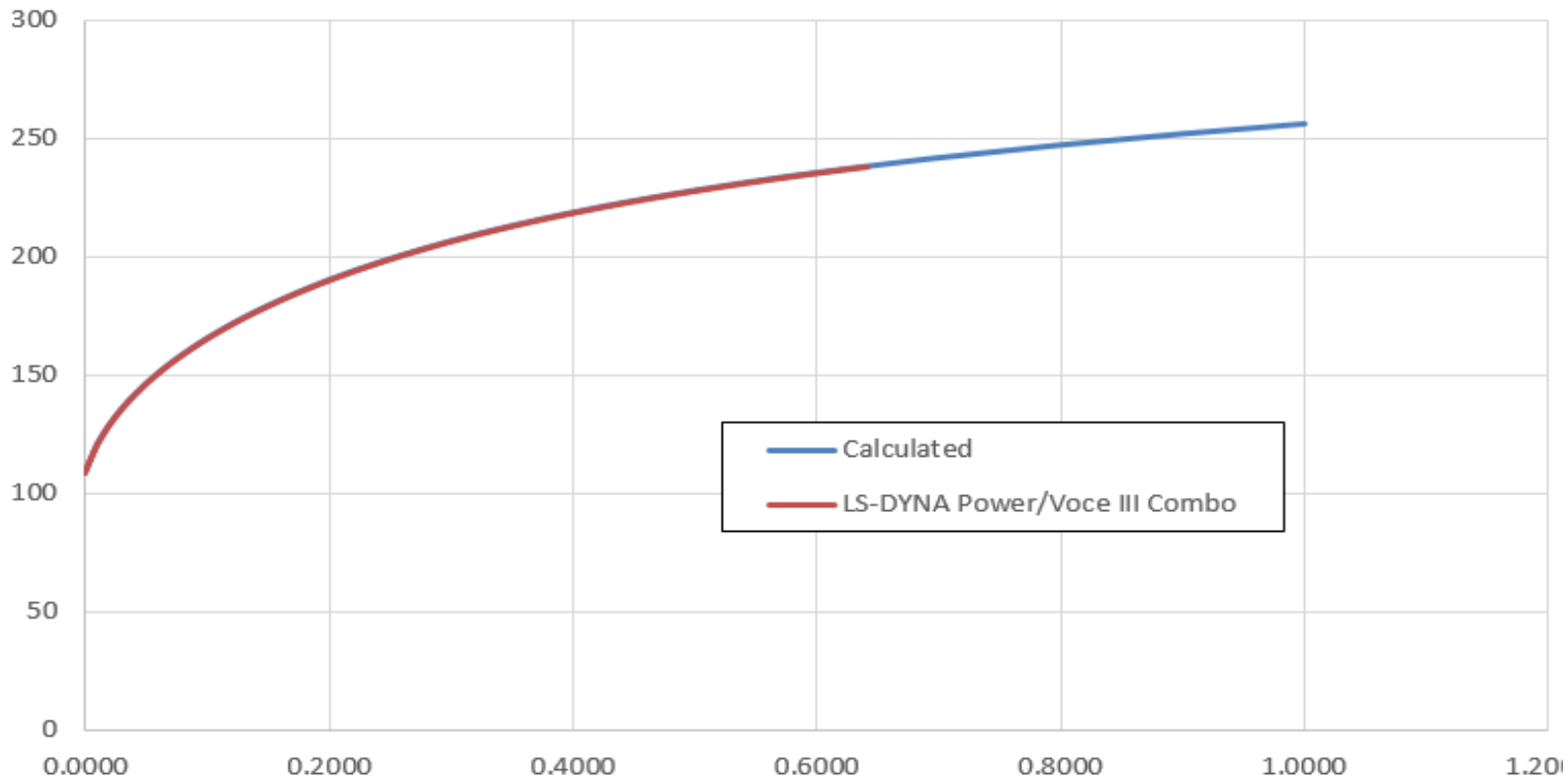
- Swift Power law  $\sigma = K(\varepsilon_0 + \varepsilon_p)^n$
- Voce law  $\sigma = \sigma_0 + R_{sat}(1.0 - e^{-\zeta \cdot \varepsilon_p})$
- Alt. Voce law  $\sigma = A - B e^{-C \cdot \varepsilon_p}$
- Hockett-Sherby law  $\sigma = A - B e^{-C \cdot \varepsilon_p^H}$
- Stoughton-Yoon law  $\sigma = A - B e^{-C \cdot \varepsilon_p^m} + D \cdot \varepsilon_p$
- Weighted combination of Swift + Voce or Hockett-Sherby

# New keyword: \*DEFINE\_CURVE\_STRESS

- Example: Weighted Combination of Swift Power Law + Voce

$$\sigma = 0.5 \cdot K(\varepsilon_0 + \varepsilon_p)^n + 0.5 \cdot (A - B e^{-C\varepsilon_p})$$

$K = 350$   
 $\varepsilon_0 = 0.01$   
 $n = 0.22$   
 $A = 162$   
 $B = 72$   
 $C = 4.3$





**Best-fit for comparison of  
geometrical deviations**



## \*CONTROL\_FORMING\_BESTFIT

- Comparison of geometries from simulation and scan of real part.
  
- Rigidly moves a source (mesh) to the target (e.g. STL) so they maximally coincide:
  - Source mesh: predicted springback shape (e.g. dynain file)
  - Target mesh: scan data (e.g. stl file)
  - Result file: bestfit.out (dynain file with shell thickness as normal deviations between the two parts)
  
- Limitations:
  - shell elements only
  - double precision only.
  
- Available now in LS-PrePost 4.3 in Metal Forming Application/eZ Setup.

# LS-PrePost interface for \*CONTROL\_FORMING\_BESTFIT

The image shows the LS-PrePost software interface. On the left, the 'Application' menu is open, with 'Metal Forming' selected. The 'Best Fit' option is highlighted in the sub-menu. On the right, the 'Best Fit' dialog box is open, showing the 'By LS-DYNA' radio button selected. The 'Target Part' is set to 1, and the 'Source Part' is set to 2. The '3 Nodes on Target' and '3 Nodes on Source' checkboxes are unchecked. The 'Output' button is highlighted.

1: Application menu  
2: Metal Forming  
3: Best Fit  
4: By LS-DYNA radio button  
5: Target Part checkbox  
6: Source Part checkbox  
7: 3 Nodes on Target value  
8: 3 Nodes on Source value  
9: Output button

10: Run LS-DYNA double precision.

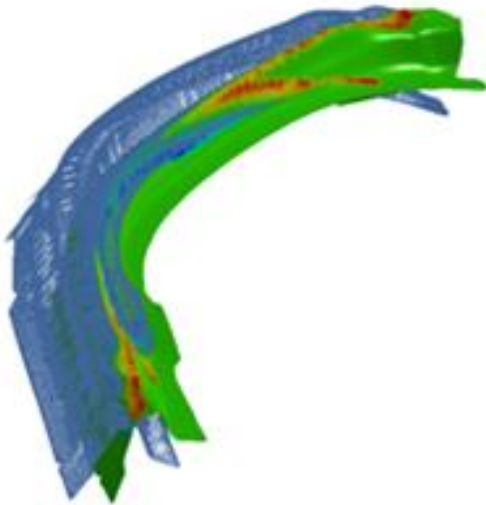
# Example for \*CONTROL\_FORMING\_BESTFIT

Before best fit

After best fit

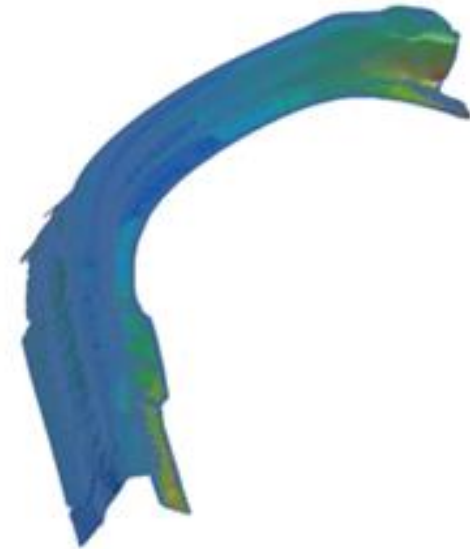
min=-14.7804, at elem# 283482  
max=14.6371, at elem# 282332

1.464e+01  
1.170e+01  
8.780e+00  
5.821e+00  
2.883e+00  
-5.813e-02  
-2.995e+00  
-5.833e+00  
-8.872e+00  
-1.181e+01  
-1.475e+01



min=-0.925148, at elem# 321167  
max=1.10289, at elem# 318630

1.103e+00  
8.001e-01  
6.973e-01  
4.945e-01  
2.917e-01  
8.887e-02  
-1.130e-01  
-3.167e-01  
-5.195e-01  
-7.223e-01  
-9.251e-01



Deviation

Before: (-14.75, 14.64)

After: (-0.925148, 1.10289)



**Improvements for One-Step solver**

## \*CONTROL\_FORMING\_ONESTEP\_ *OPTION*

### ■ \*CONTROL\_FORMING\_ONESTEP\_ **TRIA**

- activates original implementation from 2011
- all quadrilateral elements are internally split into 2 triangular elements

### ■ \*CONTROL\_FORMING\_ONESTEP\_ **QUAD**

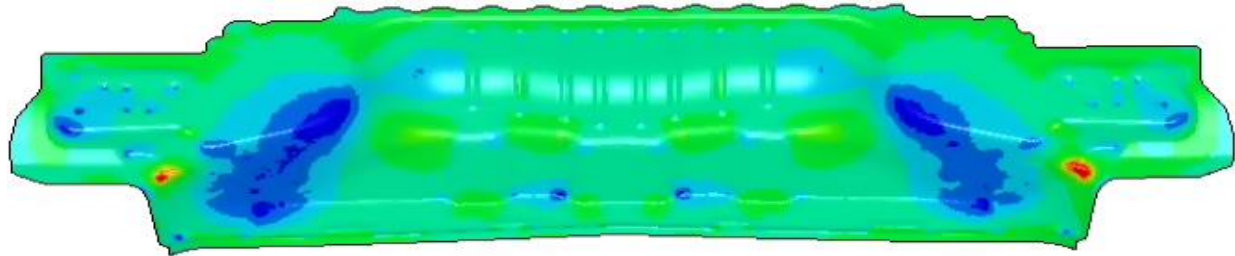
- quadrilateral elements with improved algorithm
- better results
- Improved calculation speed on multiple CPUs/Cores (SMP)

### ■ \*CONTROL\_FORMING\_ONESTEP\_ **QUAD2**

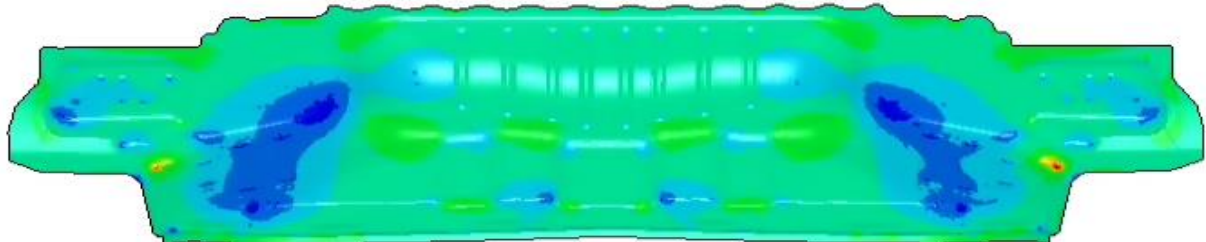
- Further improved element formulation but slightly longer CPU times than option QUAD
- Better prediction of thinning and plastic strains
- New default

# \*CONTROL\_FORMING\_ONESTEP\_OPTION

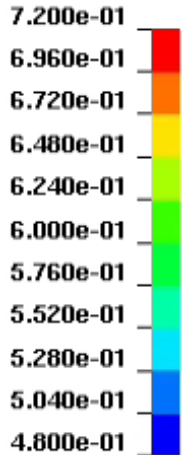
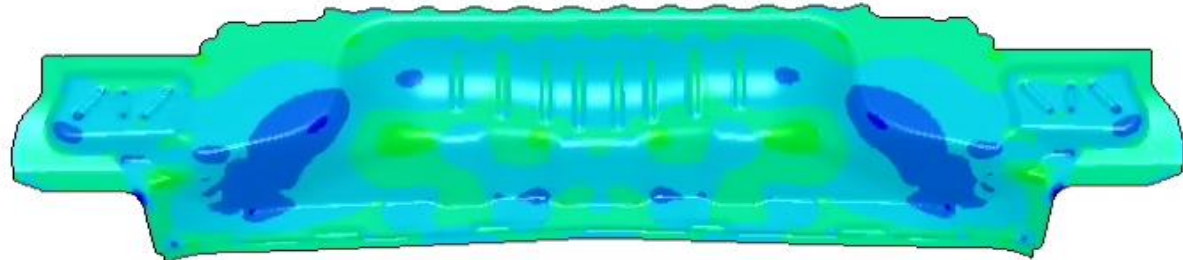
■ TRIA



■ QUAD



■ QUAD2



	Number of elements	Calculation speed (D.P. SMP Rev.112720, 8 CPUs)		
		Option TRIA	Option QUAD	Option QUAD2
A hat shape part	71000	21.0 min	14.1 min	16.6 min
A upper dash panel	61700	24.5 min	11.5 min	17.2 min

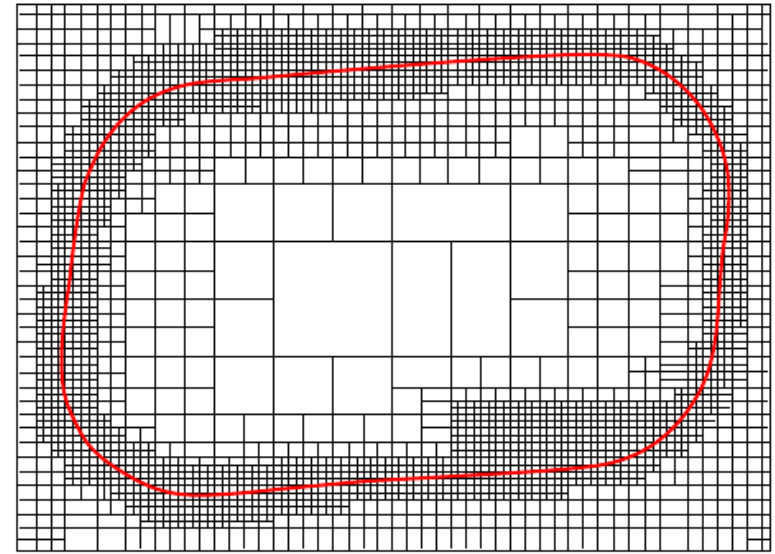
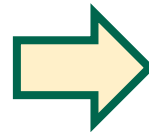
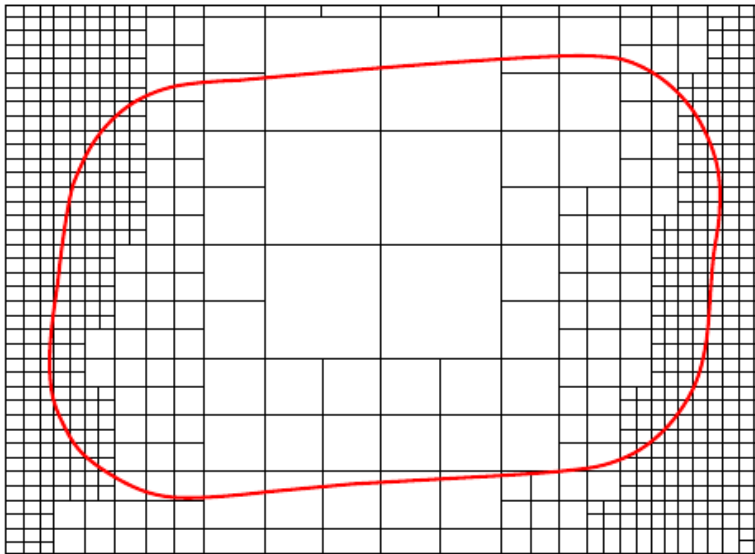


**Uniform mesh refinement of  
selected region**

## \*CONTROL\_ADAPTIVE\_CURVE

Card 1	1	2	3	4	5	6	7	8
Variable	IDSET	ITYPE	N	SMIN	ITRIOPT			
Type	I	I	I	F	I			

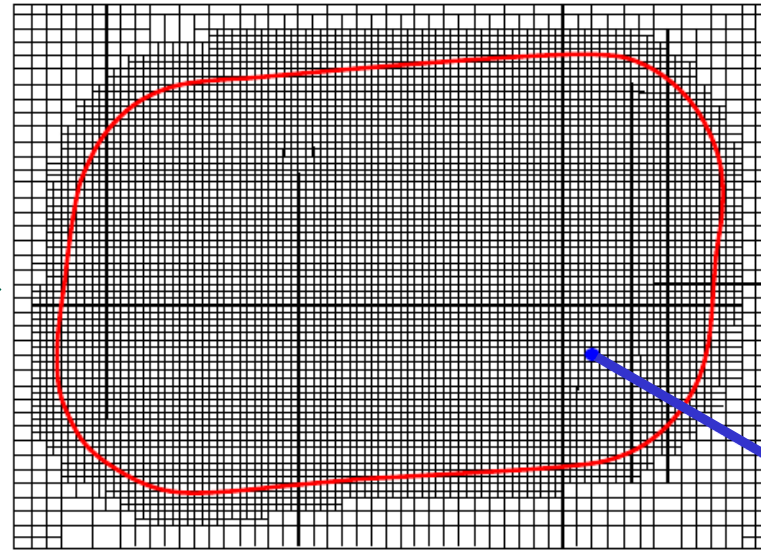
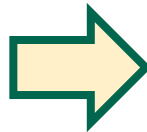
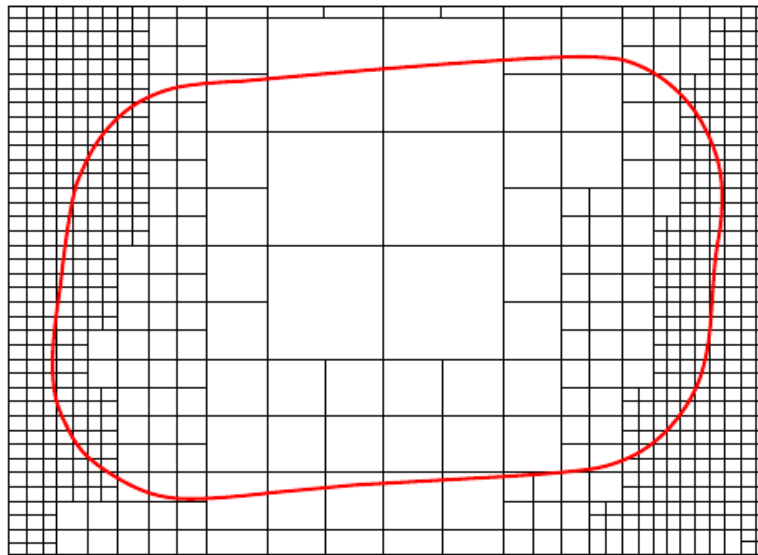
- ITRIOPT=0 → Refine shell elements along a curve





# \*CONTROL\_ADAPTIVE\_CURVE

- ITRIOPT=1 → Refine shell elements along and inside a curve loop
  - Seed node has to be defined



Seed node

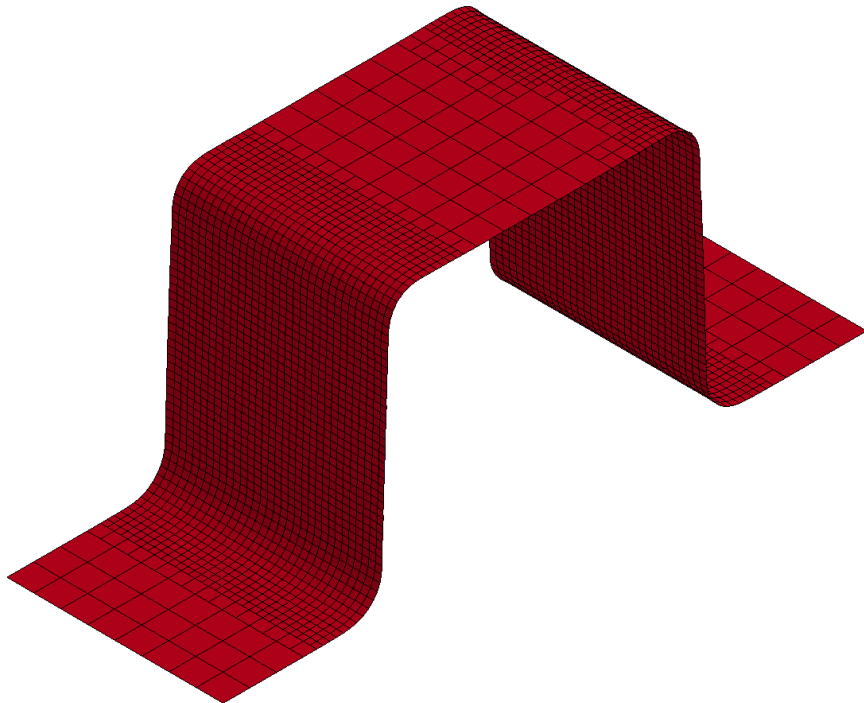


# Mesh fusion

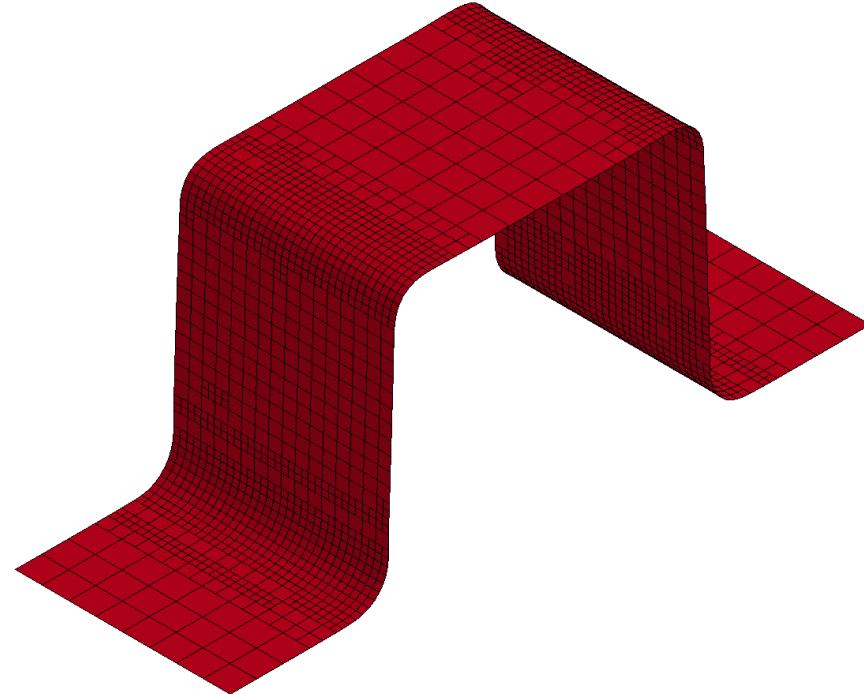
# Mesh fusion

- Mesh fusion so far was available for **SMP** only
- Now mesh fusion also is available for **MPP**
- Fusion can be configured via `*CONTROL_ADAPTIVE`

without fusion

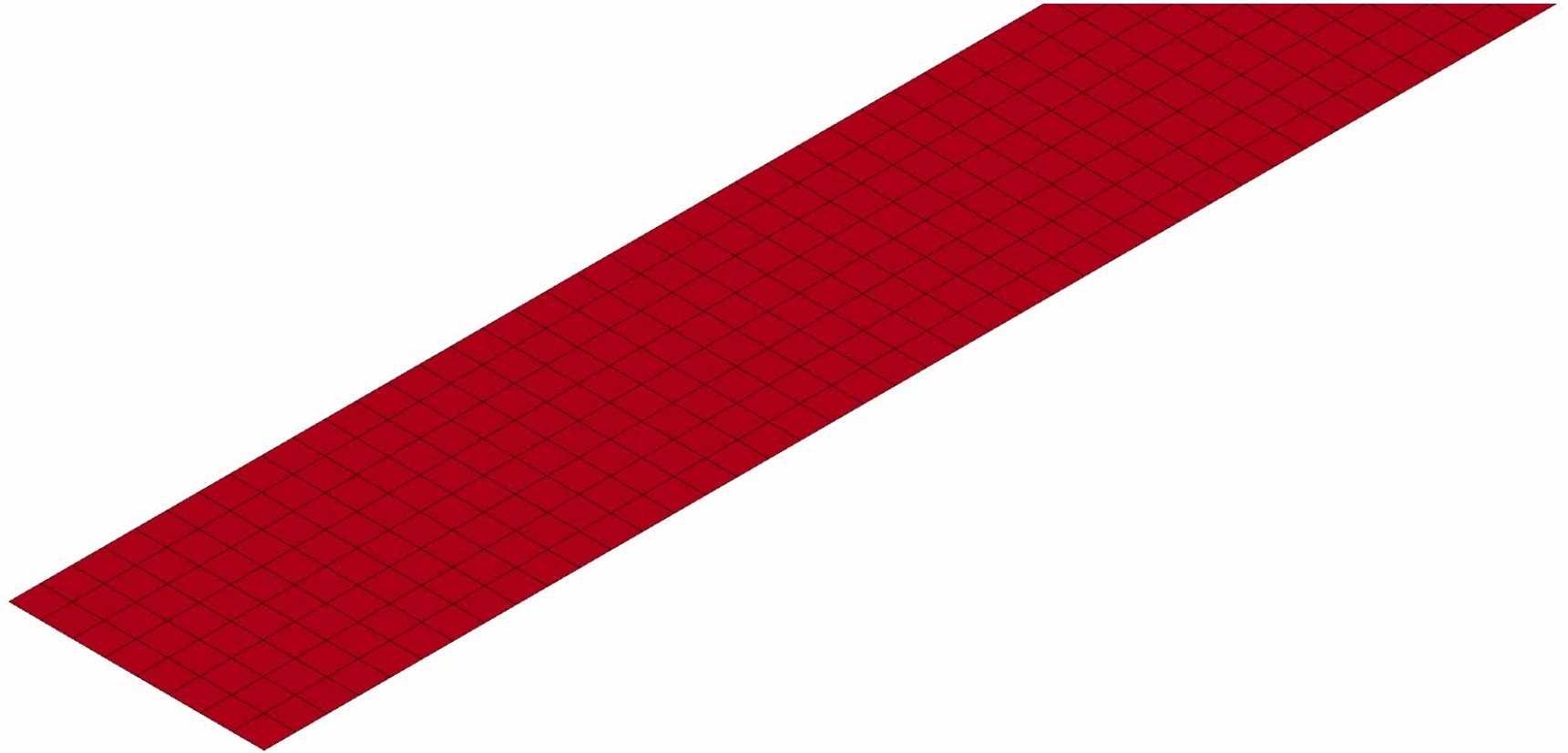


with mesh fusion





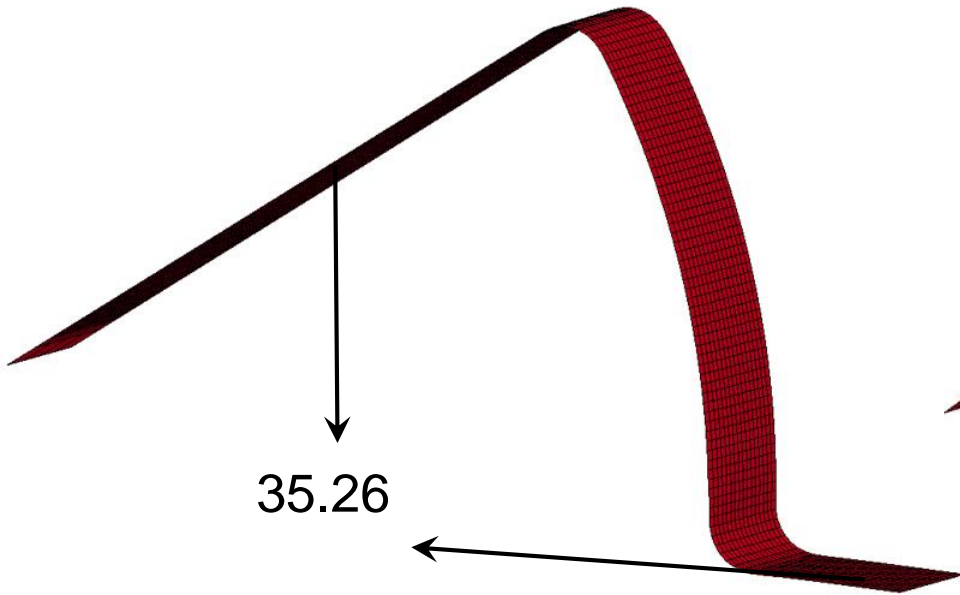
# Mesh fusion



# Mesh fusion

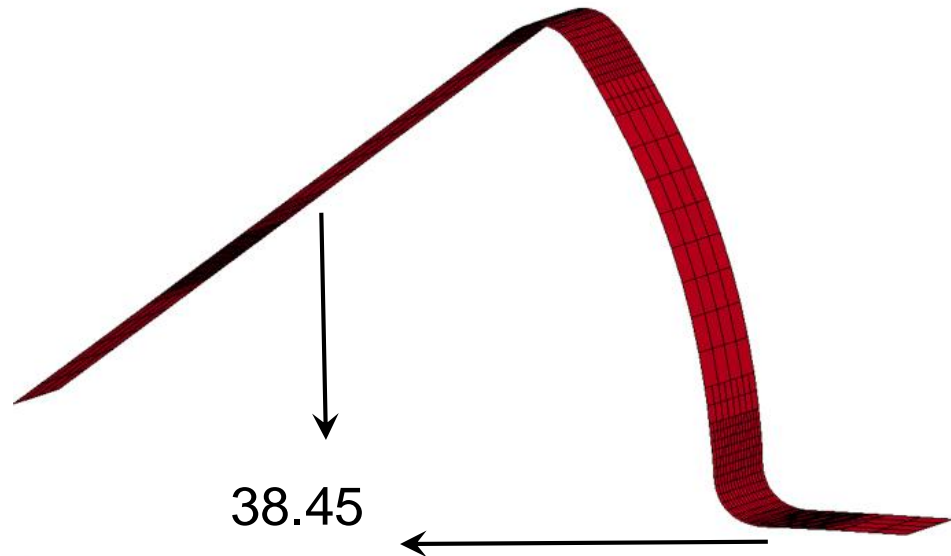


Springback **without** fusion



Final number of shell elements = **2476**

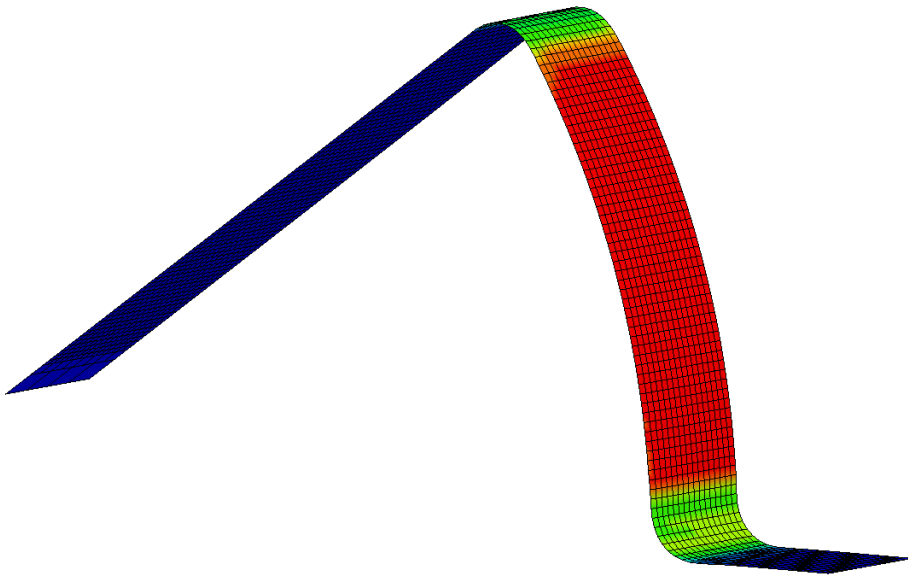
Springback **with** mesh fusion



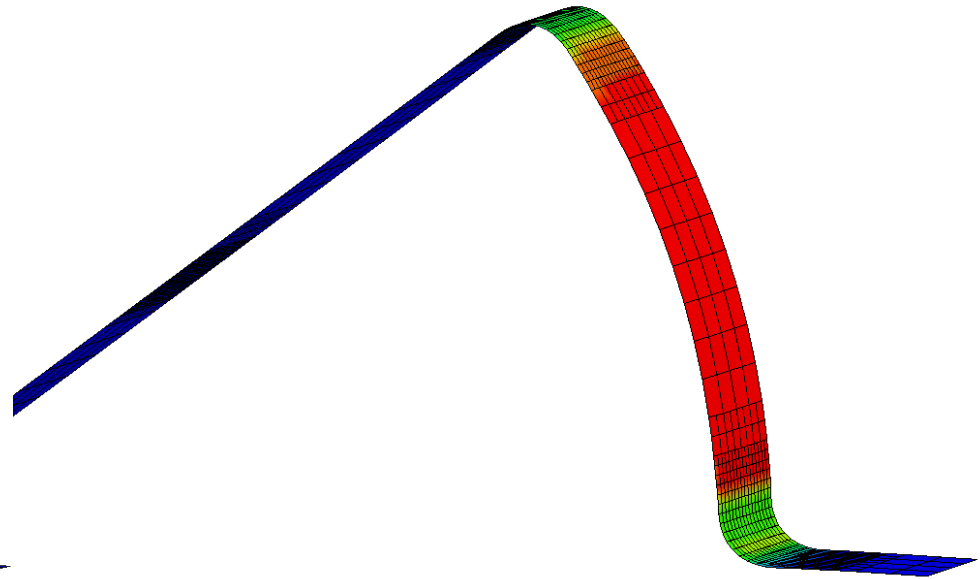
Final number of shell elements = **1000**

# Mesh fusion

effective plastic strain  
**without** fusion

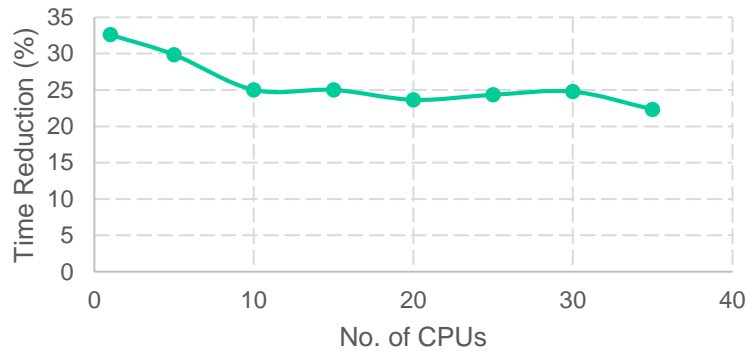


effective plastic strain  
**with** mesh fusion

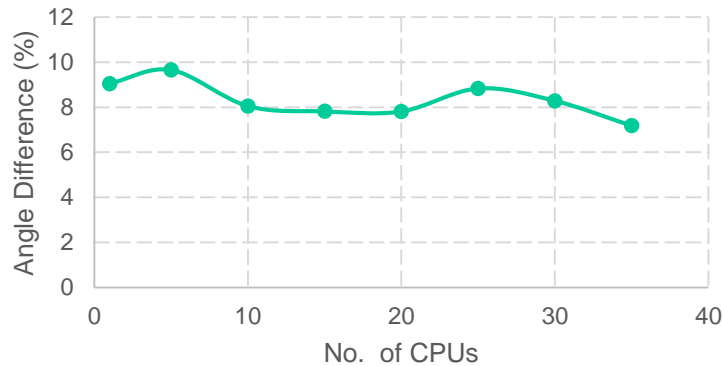


# Mesh fusion

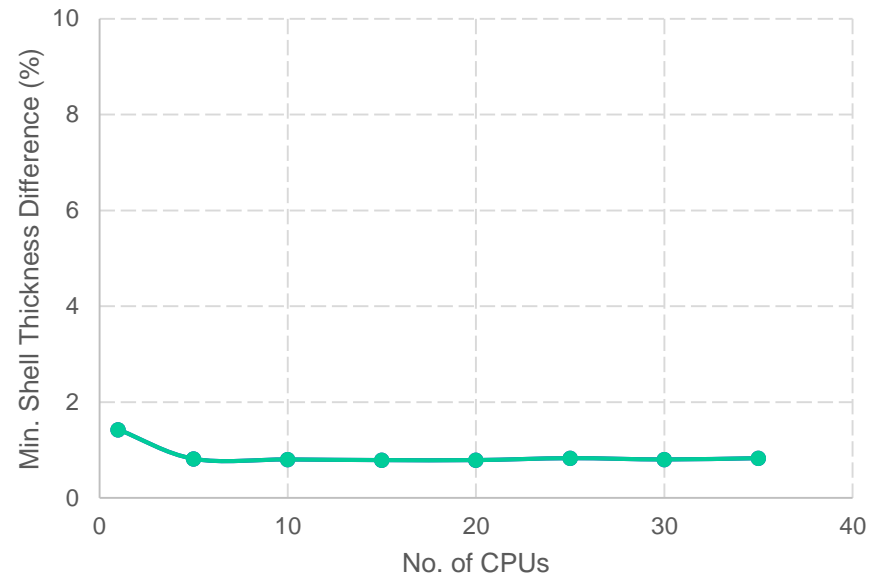
## Time Reduction



## Springback Angle Difference



## Min. Shell Thickness Difference



- Significant reduction in CPU cost
- No obvious difference in thinning and plastic strain prediction

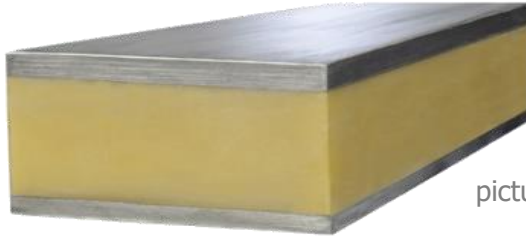


# **Adaptivity and Trimming for sandwich sheets**



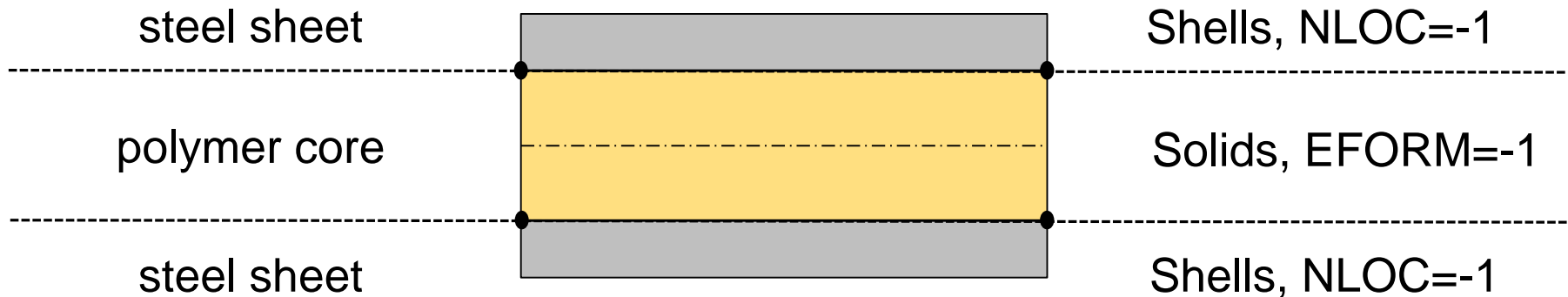
# Sandwich sheets

- Sandwich: top and bottom steel sheets and a polymer core



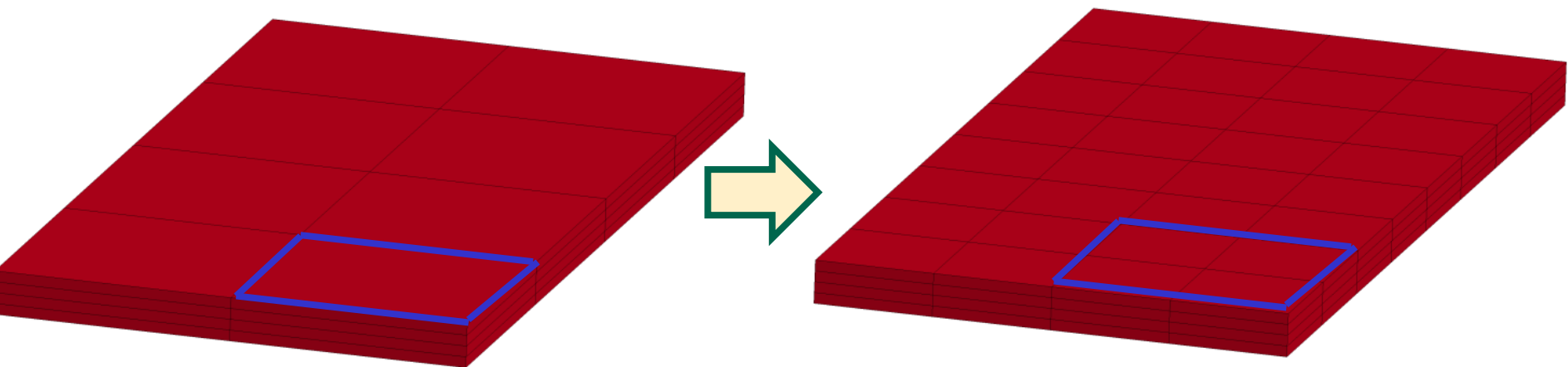
picture credit:  
Wikipedia

- Modeling of sandwich sheets:
  - Top and bottom steel sheets: shell elements
  - Polymer core: solid elements

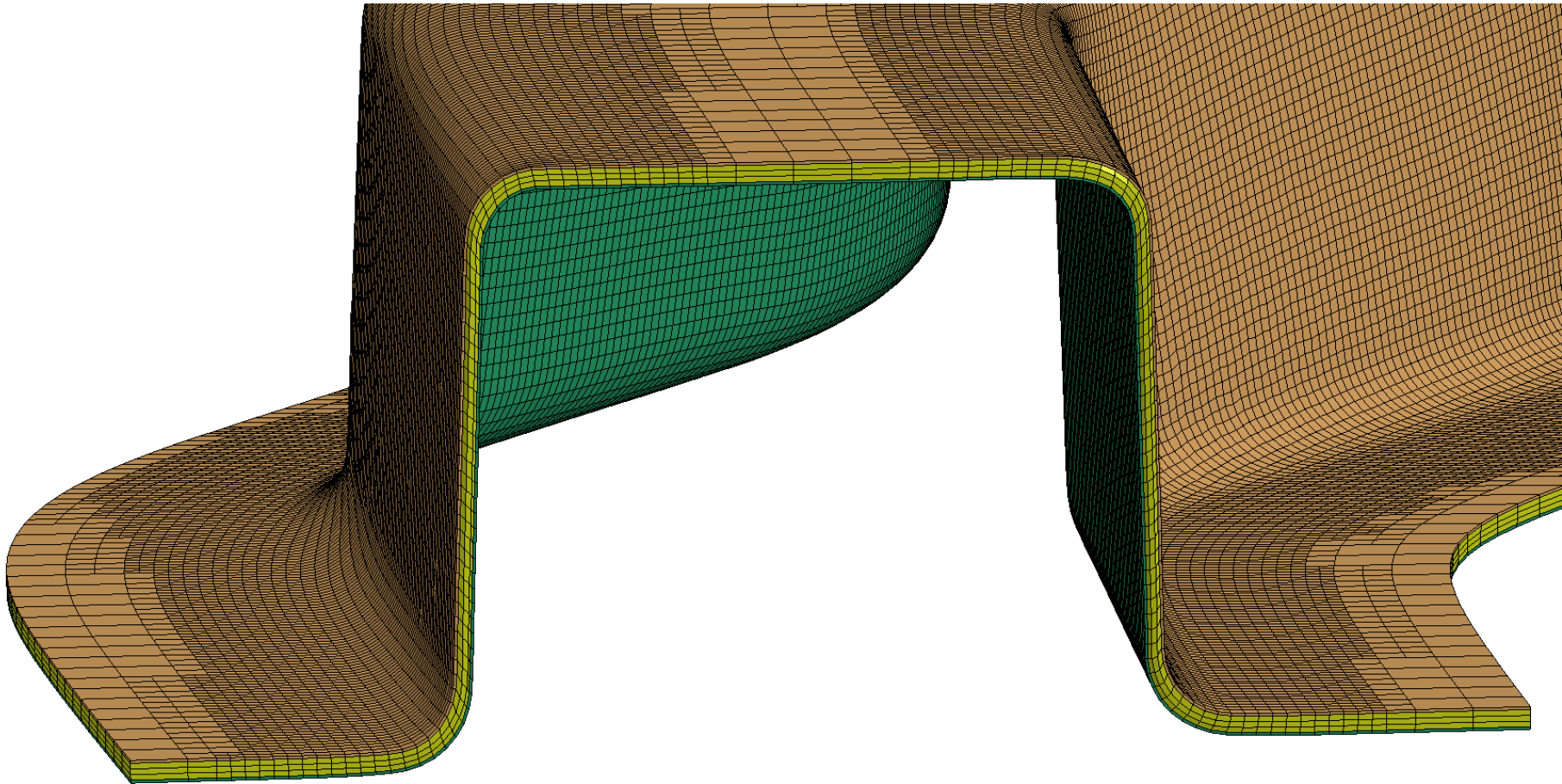


# Adaptivity for sandwich sheets

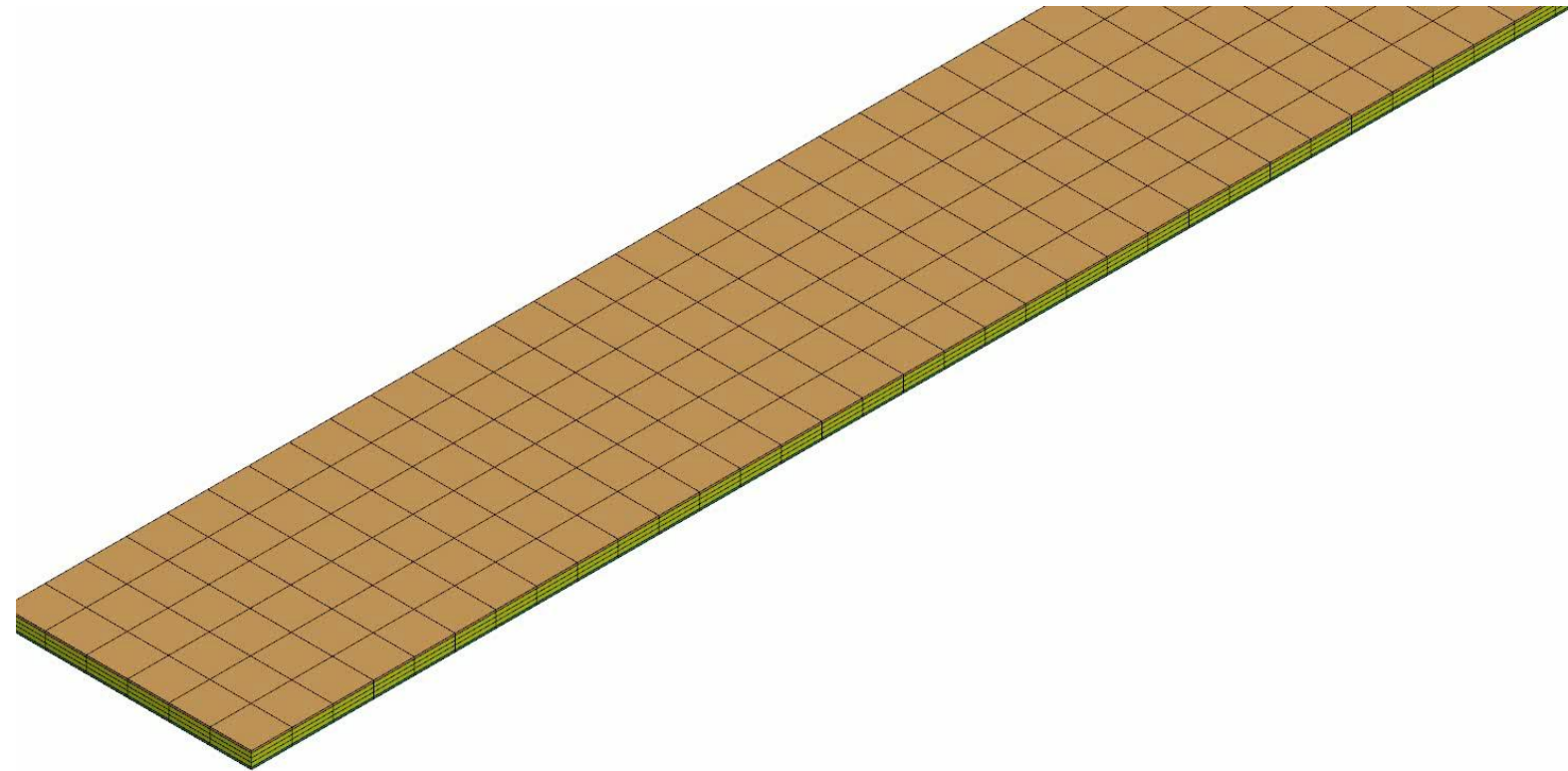
- Option **IFSAND=1** on **\*CONTROL\_ADAPTIVE** for adaptive refinement
- Mesh refinement only can happen in the plane of the sheet
- No refinement in the thickness direction
- One solid element is split into 4 solid elements
- Recent improvements:
  - Adaptivity so far was only possible for one solid element over thickness of core
  - Recently adaptivity was extended to multi-layer of solid elements
- Works for both SMP and MPP



# Adaptivity for sandwich sheets

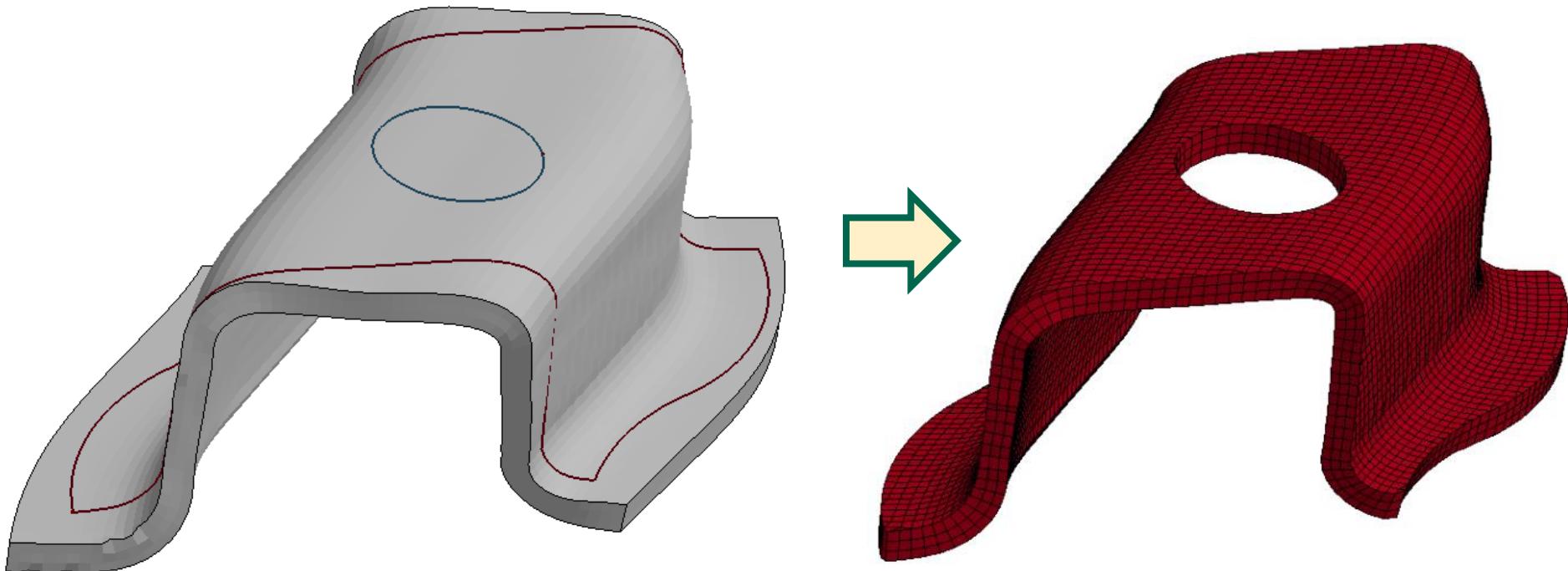
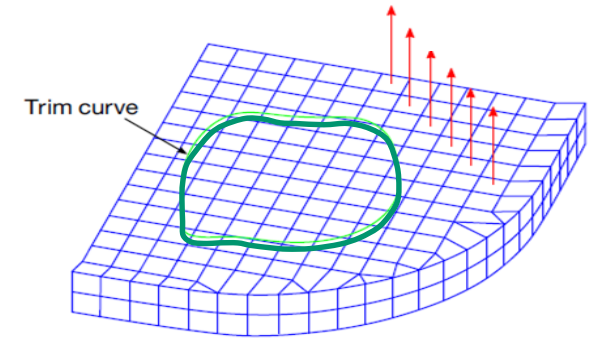


# Adaptivity for sandwich sheets



# Trimming of sandwich sheets

- 2D and 3D trimming of sandwich sheet and solids
- Definition is similar as for trimming of shell elements
- Additional input to indicate solid normals: TDIR on `*DEFINE_CURVE_TRIM_3D`





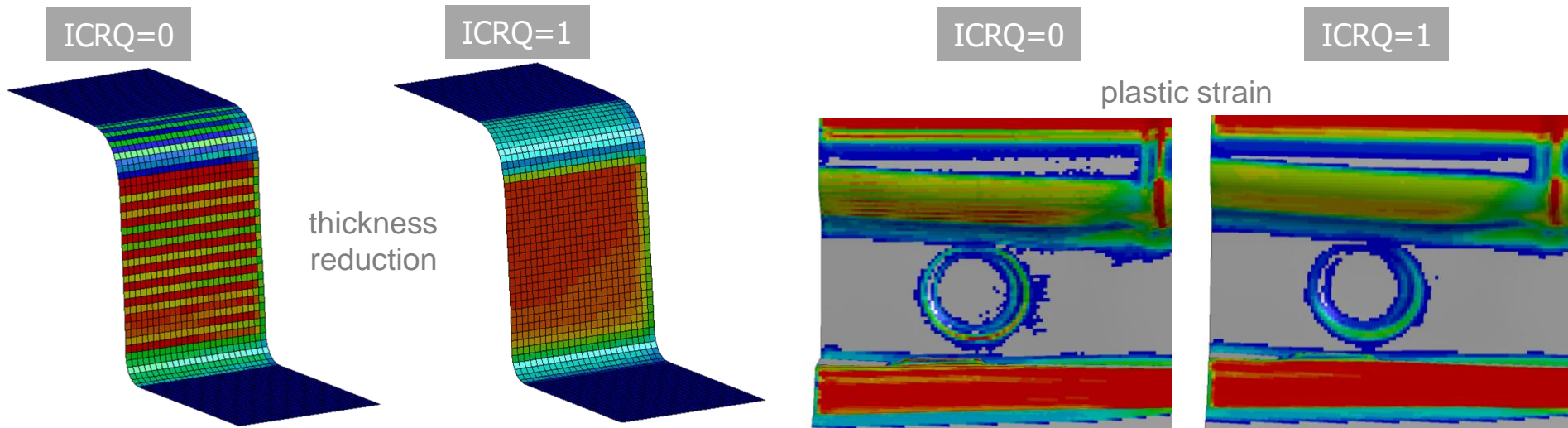
**Miscellaneous**



# Continuous result quantities

## ■ New option ICRQ on \*CONTROL\_SHELL

- Continuous treatment of thickness and plastic strain across element edges for shell element formulations 2, 4, and 16 with max. 9 integration points through the thickness
- Reduces alternating weak localizations sometimes observed in metal forming applications when shell elements get stretch-bended over small radii
- Similar to MAT\_NONLOCAL



# Miscellaneous

- Improvements to **\*CONTROL\_FORMING\_AUTOCHECK**
  - E.g. output rigid tool mesh in offset position
- **\*BOUNDARY\_SPC\_SYMMETRY\_PLANE\_SET**
  - Possibility to define symmetry constraints for part set, e.g. tailor welded blanks
- **\*CONTROL\_FORMING\_SHELL\_TO\_TSHELL**
  - Automatically change from shell to thick shell elements
- New option **\*INTERFACE\_SPRINGBACK\_EXCLUDE** to exclude selected portions from the generated dynain file





# Thank you for your attention!

